

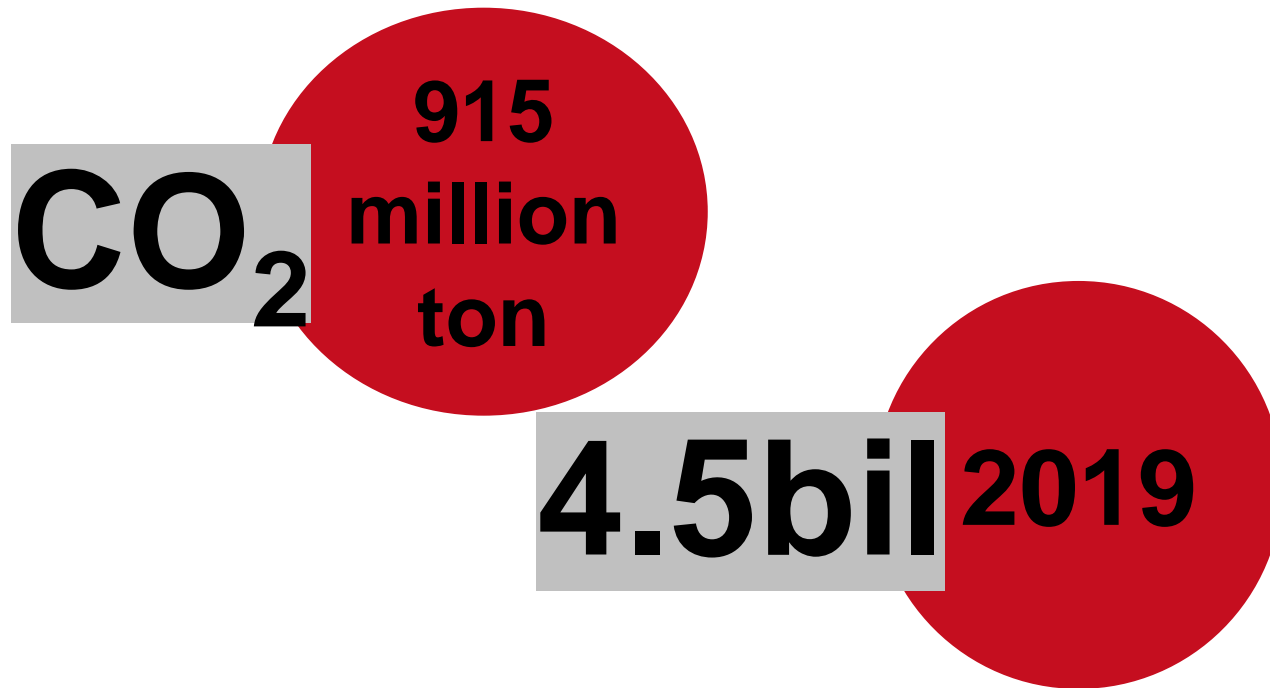


A new propulsion architecture for a hydrogen aircraft

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Problem 1: Carbon emissions



Source: atag.org



Problem 2: Long flight hours



Non-stop madness: How long-haul flights affect your mind

The world's longest non-stop flights



Perth – London
17h 20m



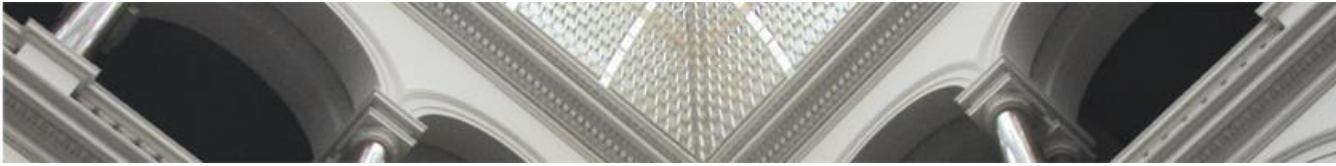
Houston – Sydney
17h 30m

[1]



A321 XLR

Flying 11 hours (4,700nm) non-stop [2]



Requirements for 2050 zero-carbon emission



**Regional aircraft
GI = 0.35**

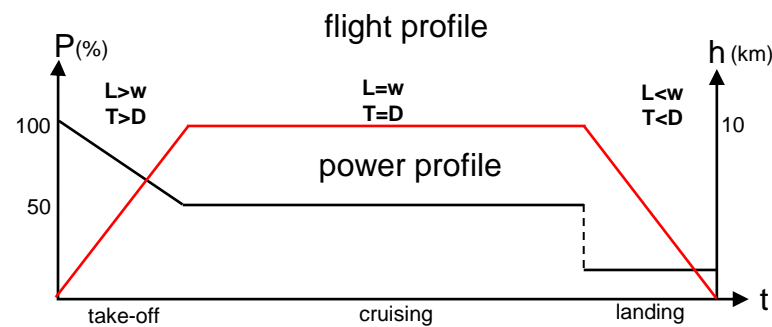
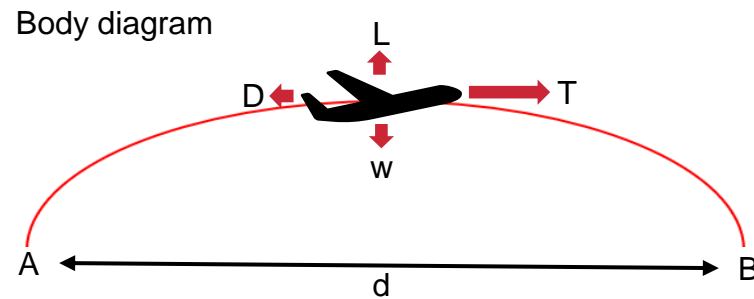
**International aircraft
GI = 0.5**

*GI = Gravimetric index

[3]



Theory and fundamental



Energy = area under the graph of power profile
Energy parameter = **thrust, speed and flight time**



Design trade-off

- Low gravimetric index
- Low boil-off
- Low speed
- High I_{sp}
- Low volume efficiency

Regional/air cargo

*International/
transatlantic aircraft*

- High gravimetric index
- High boil-off
- High speed
- Low I_{sp}
- High volume efficiency



[4]



[5]

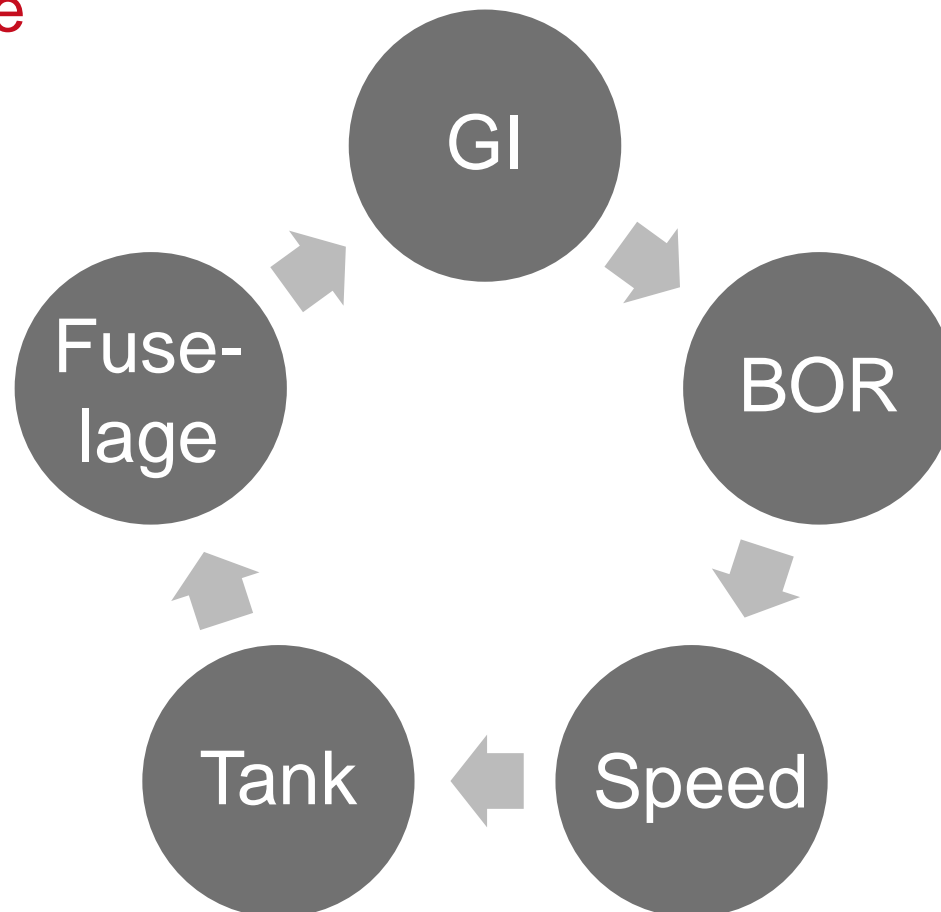


[6]



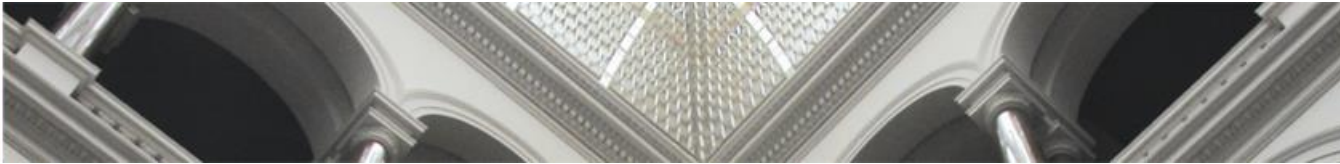
[7]

Design cycle



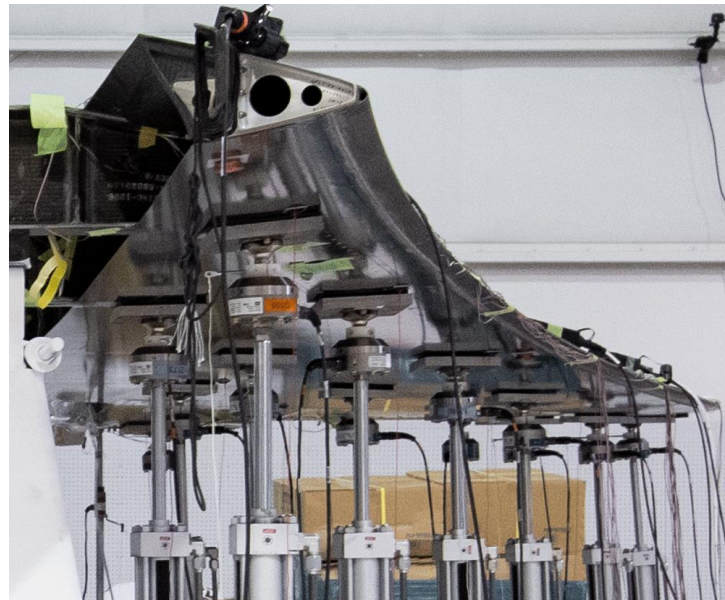
*GI = Gravimetrix index

*BOR= Boil-off rate



Design challenges

1. A high GI allows a complex geometry of a tank (e.g. wing)



[8]

*GI = Gravimetric index

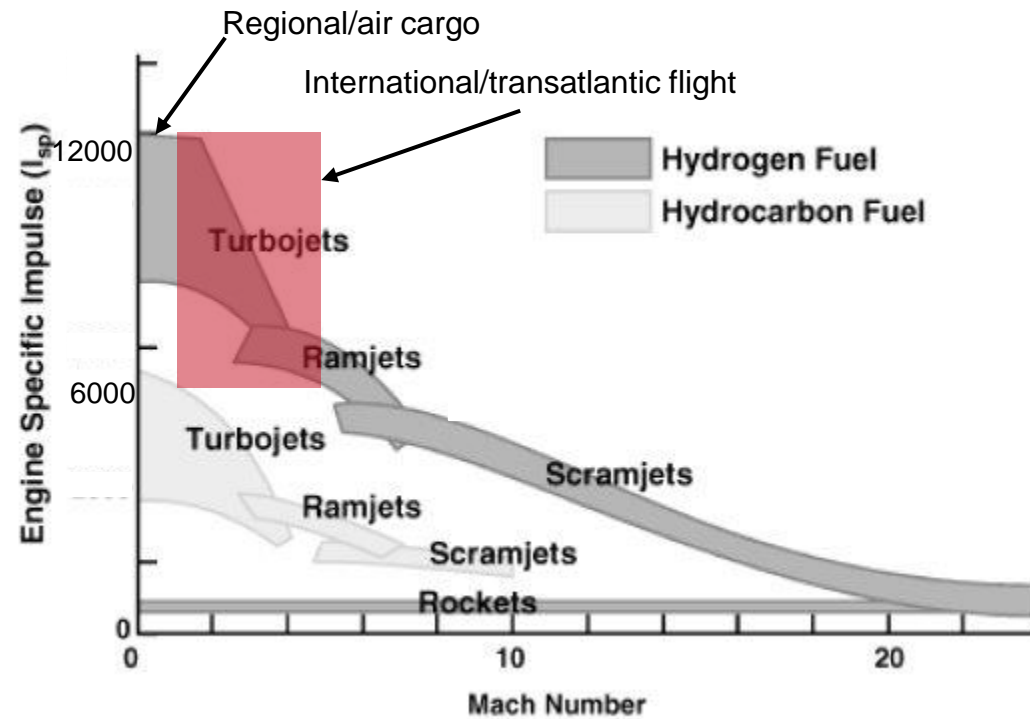
Propulsion architecture



[9]



Engine



Courtesy of the Air Force Propulsion Directorate (circa. 1990's)

[10]



LH₂ Tank

C-C

- Small
- High pressure

Li-Al

- Big
- Lightweight

Mg-Alloy

- Huge
- High strength

Thermal range



Gas turbine **3500 ±500 K**

Aero thermal 400 ±15 K

FC **310 ±10 K**

Insulator **290 ±10 K**

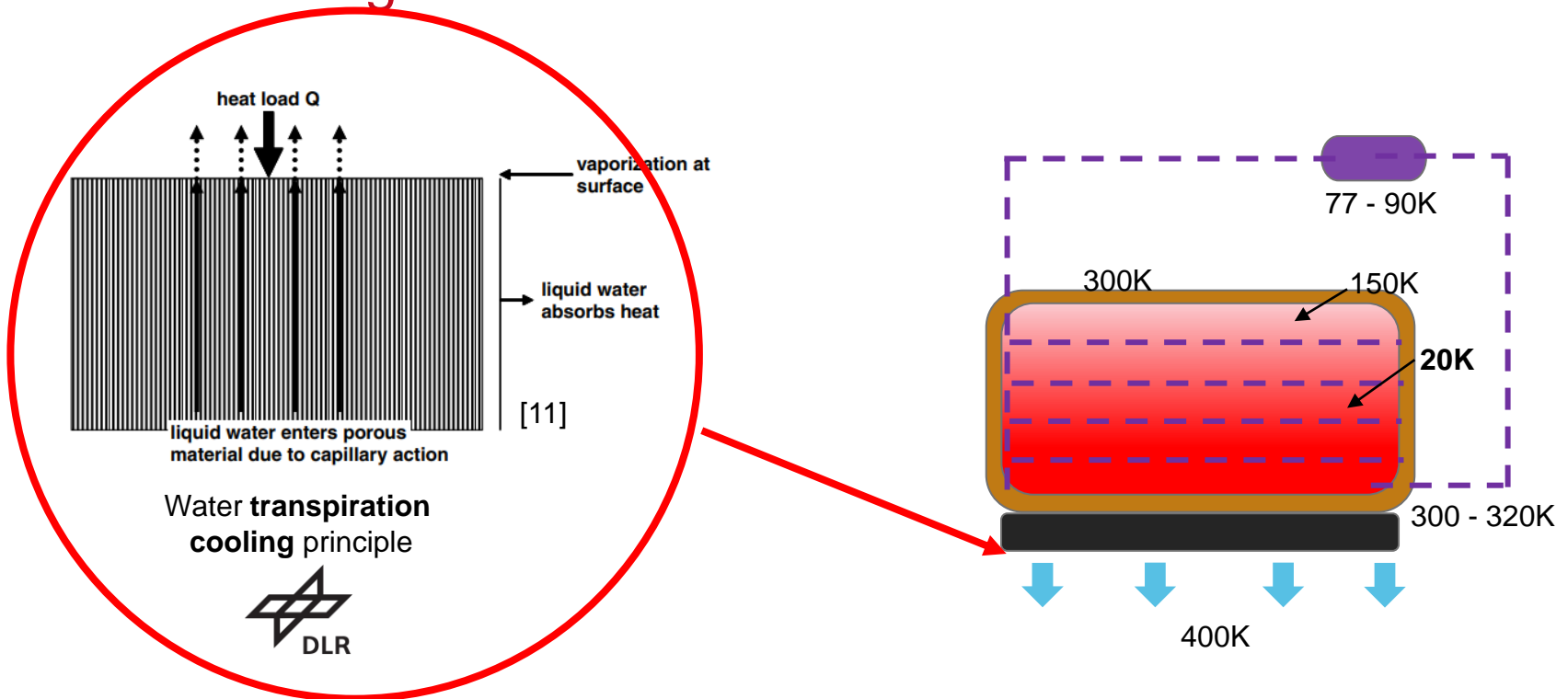
GH2 **150 ±10 K**

LN2 **77 - 90 K**

LH2 **22 ±2 K**



Thermal design





Safety: Hydrogen dumping

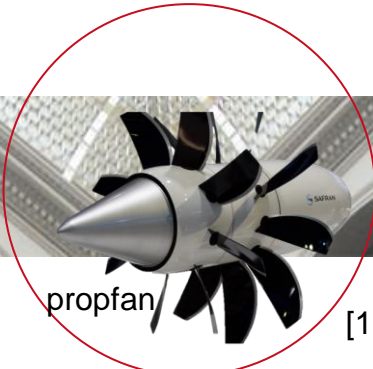
1. Emergency landing
2. Built-up pressure
3. Electrostatic charge around the aircraft body

**Combustion
w/o thrust**

**Combustion
w/o flame**

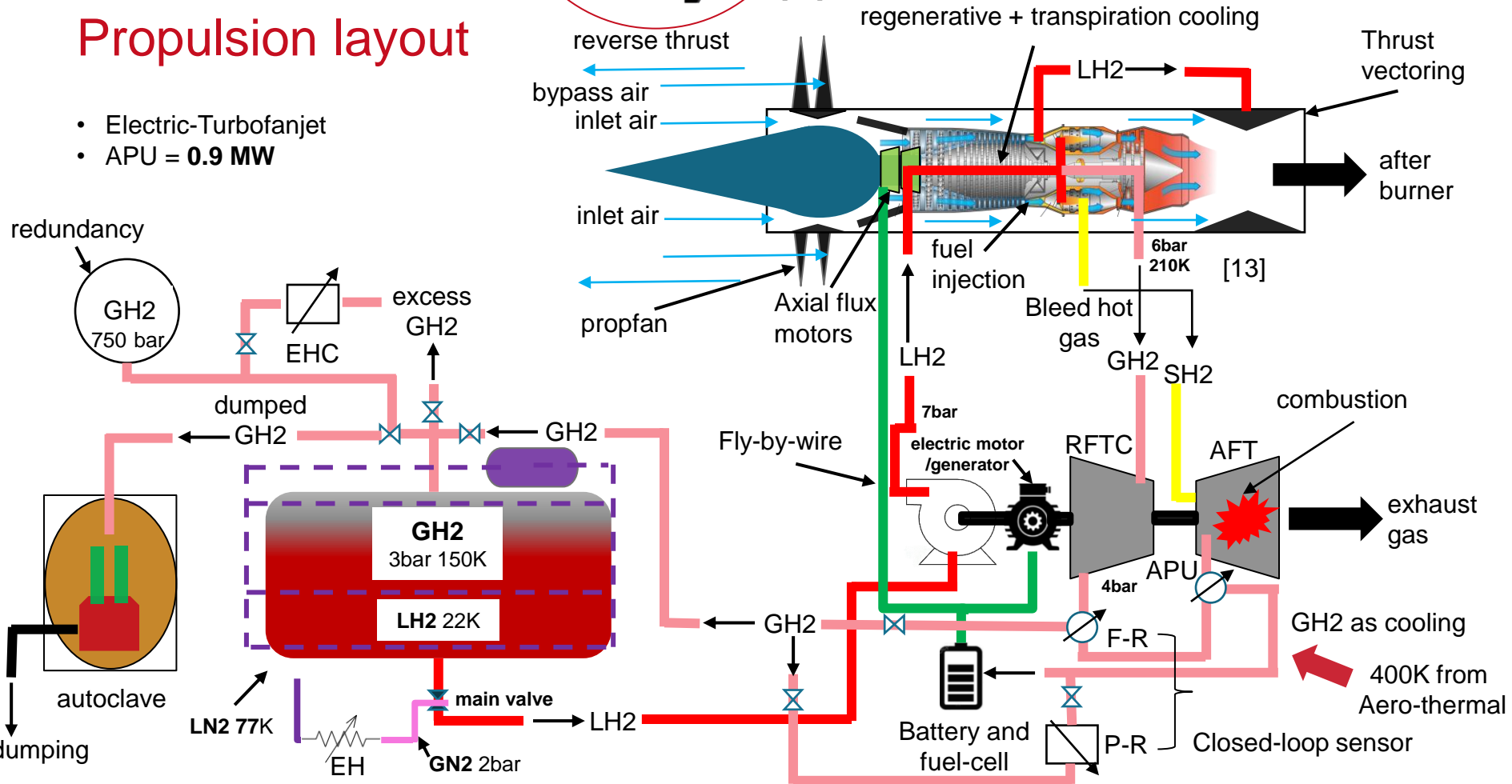
Hydrogenation

*w/o = without

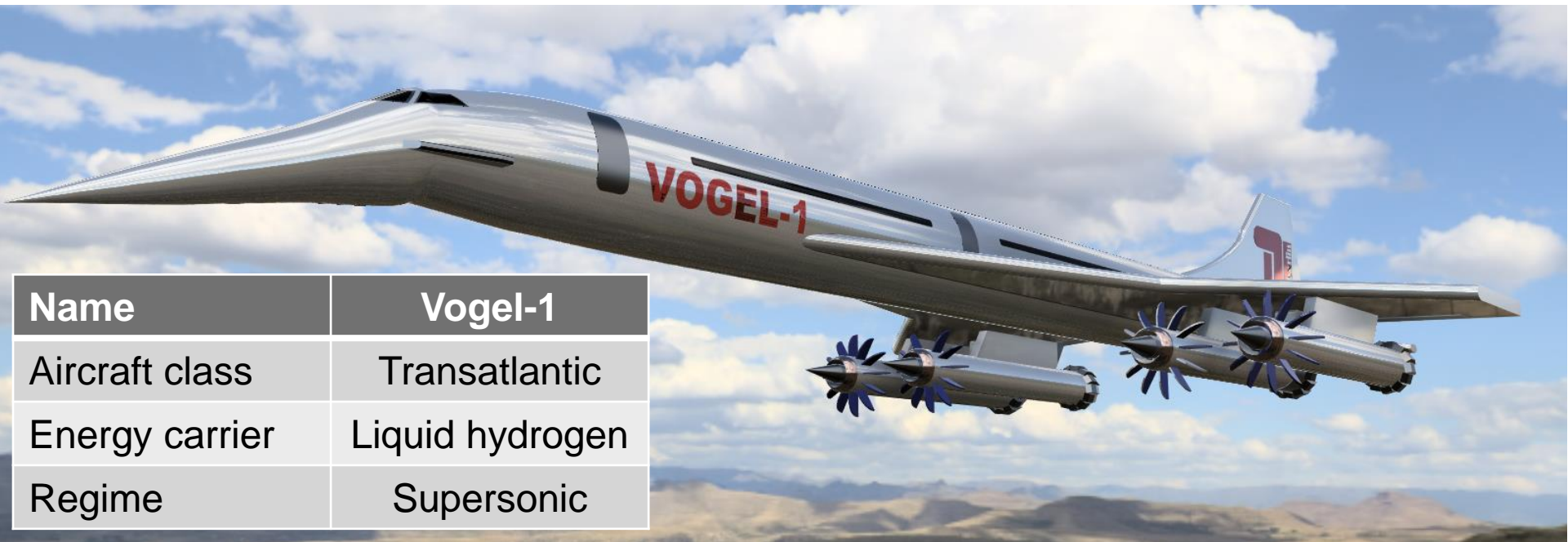


Propulsion layout

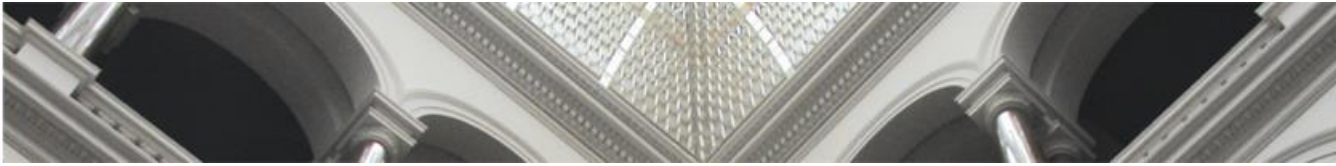
- Electric-Turbofanjet
- APU = 0.9 MW



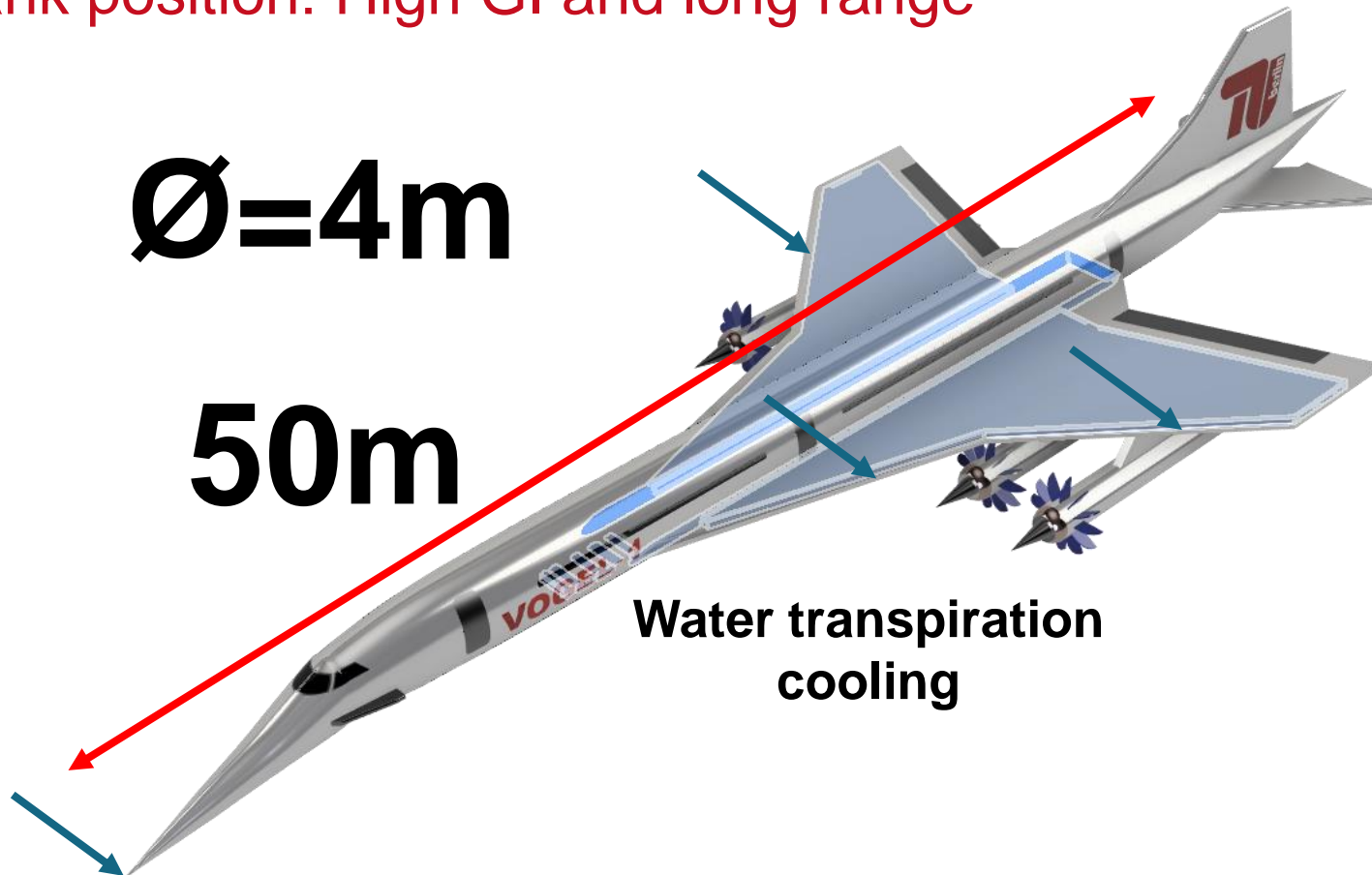
Tank position: High GI and long range



Name	Vogel-1
Aircraft class	Transatlantic
Energy carrier	Liquid hydrogen
Regime	Supersonic



Tank position: High GI and long range

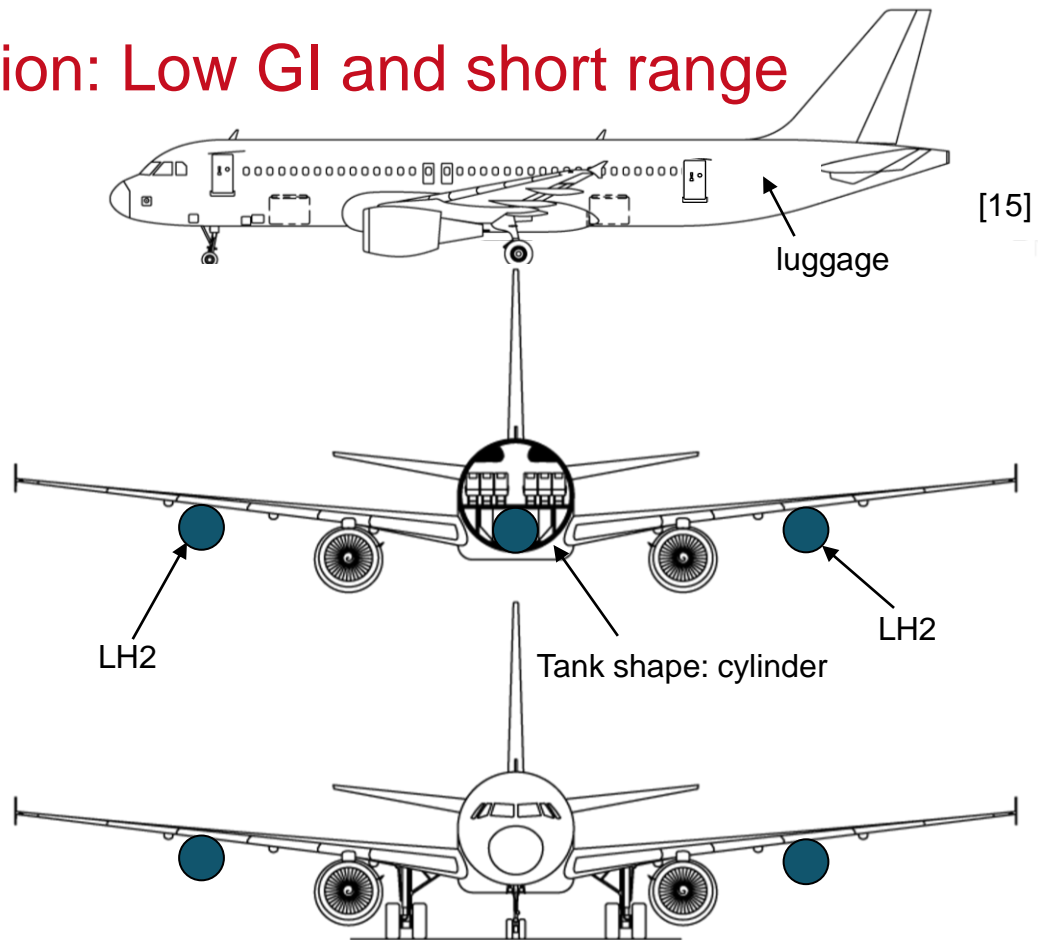




Vogel-1

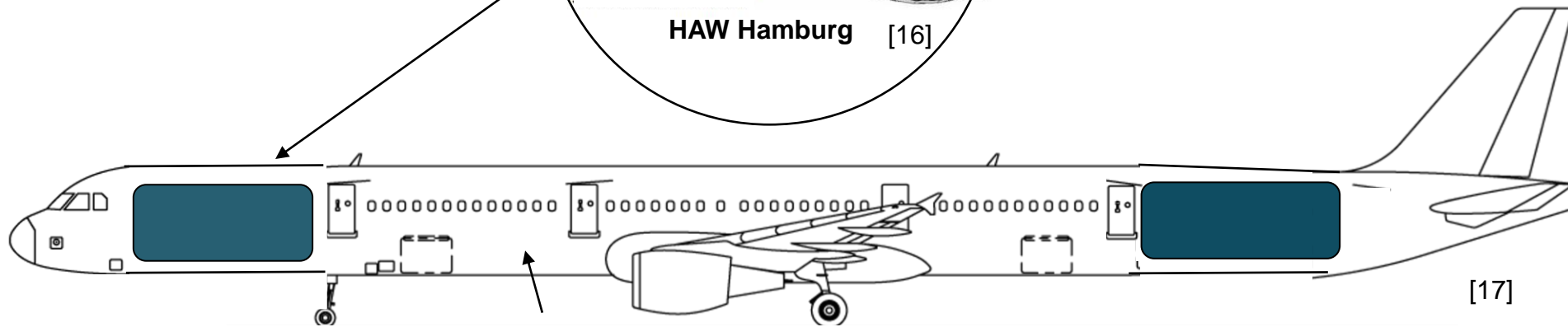
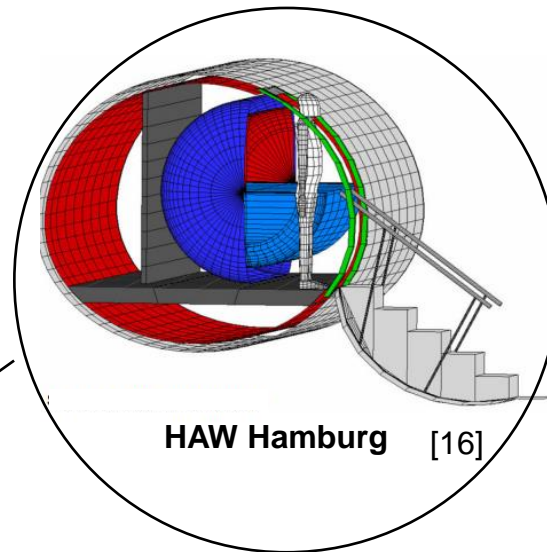
Specifications	Unit	Figures
Passenger capacity	PAX	200-240
Gravimetric index	GI	0.8
Range	km	10000
OEW	ton	63
Fuel tank	m ³	370
Heat flux (maximum)	W/m ²	150
Wing thickness	m	0.85 (average)
Wing surface area	m ²	400
Propulsion	-	hybrid-turbojet
Engine power (1unit)	MW	27.4
Engine thrust (1unit)	kN	40

Tank position: Low GI and short range



Tank position: Low GI and long range

Tank shape: Cylinder



luggage

Trade-off results

Parameter	Unit/Symbol	SAF	H ₂ sub	H ₂ super
Range	km	10000	10000	10000
Passenger	PAX	200	200	200
Grav index	GI	0.9	0.38	0.8
Glide ratio	L/D	18	18	7
Boil-off-rate	%	0.01	0.1	18.0
Speed	Mach	0.78	0.85	2.0
Fuel tank	ton	2.9	36.3	5.5
Isp	s	6040	12600	10000
Fuselage	m	44.8	58.2	50.0
Carried fuel	ton	34.7	24.3	26.5



Trade-off summary

Low GI short range

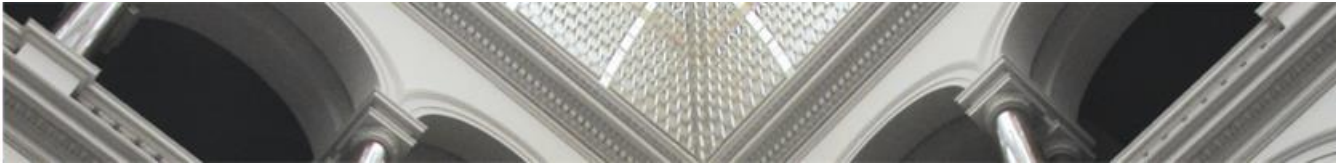
- Medium size
- Moderate flight hours
- Fuel saving

High GI long range

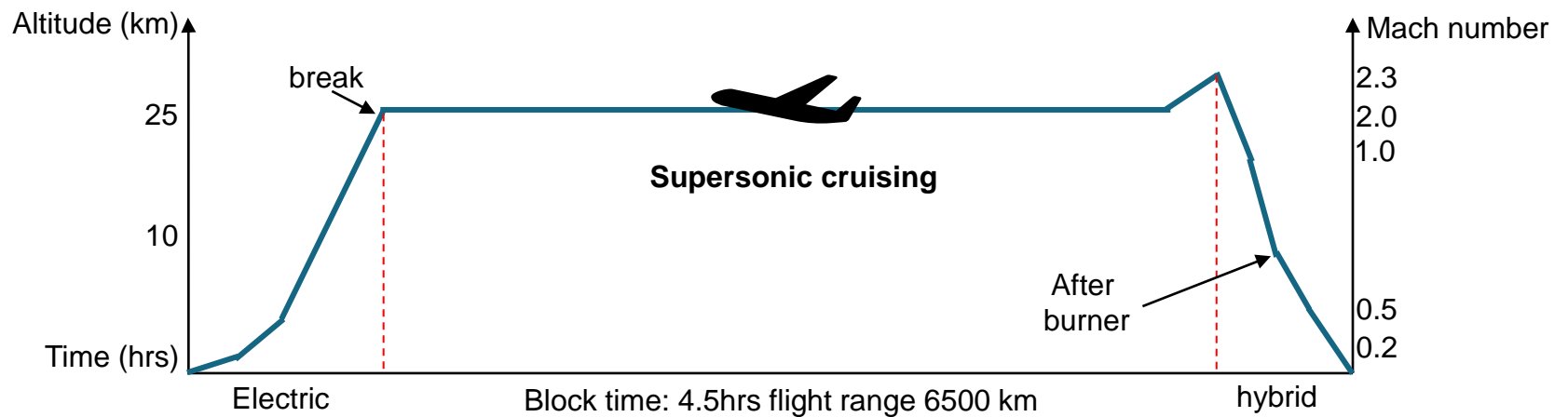
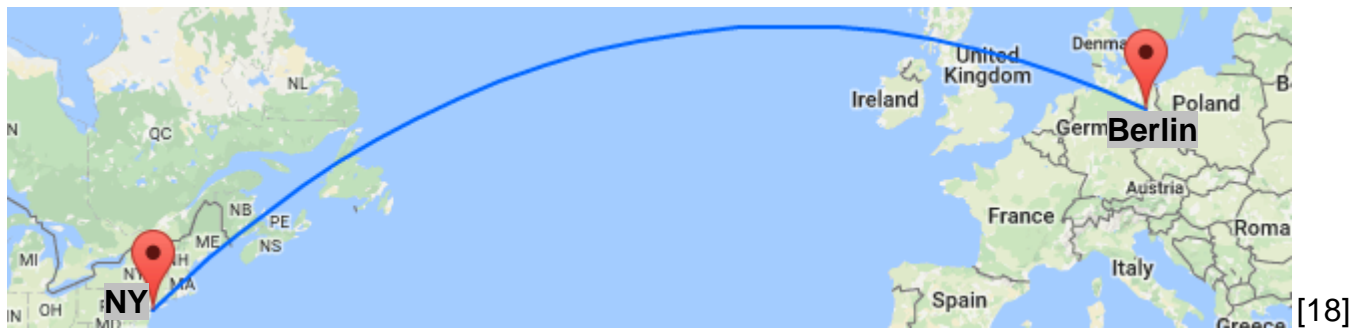
- Compact size
- Short flight hours
- Fuel economy

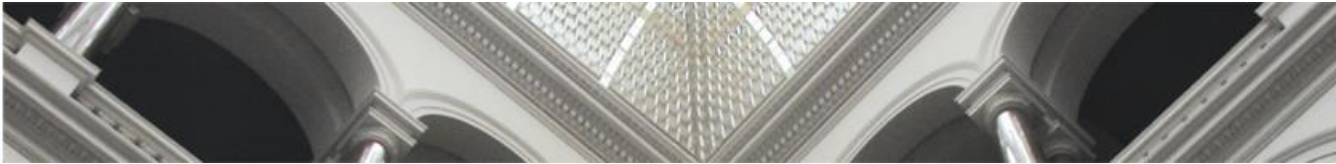
Low GI long range

- Bulky size
- Long flight hours
- Fuel saving

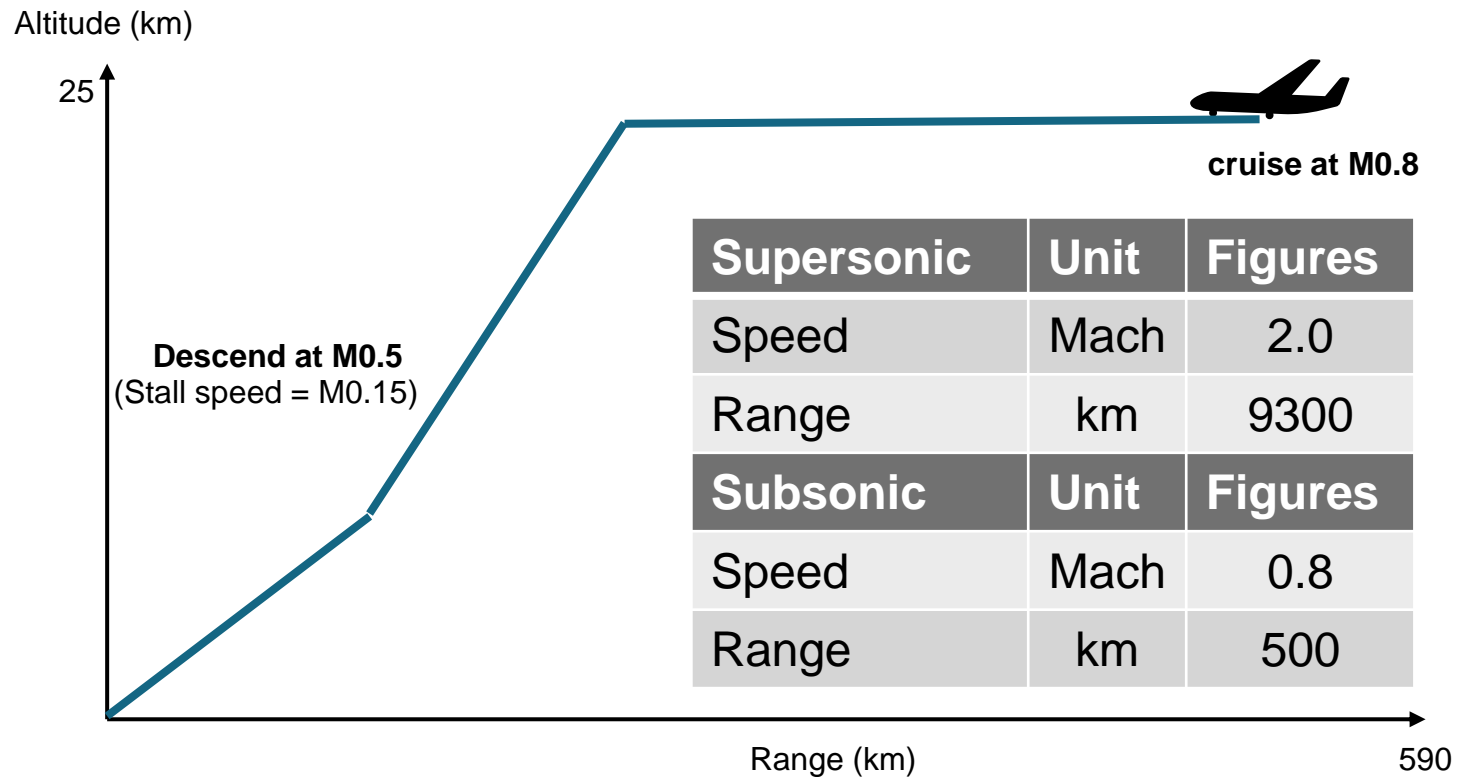


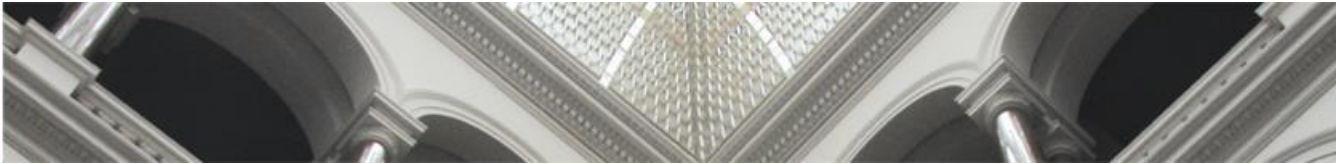
Berlin-New York





Supersonic to subsonic flight



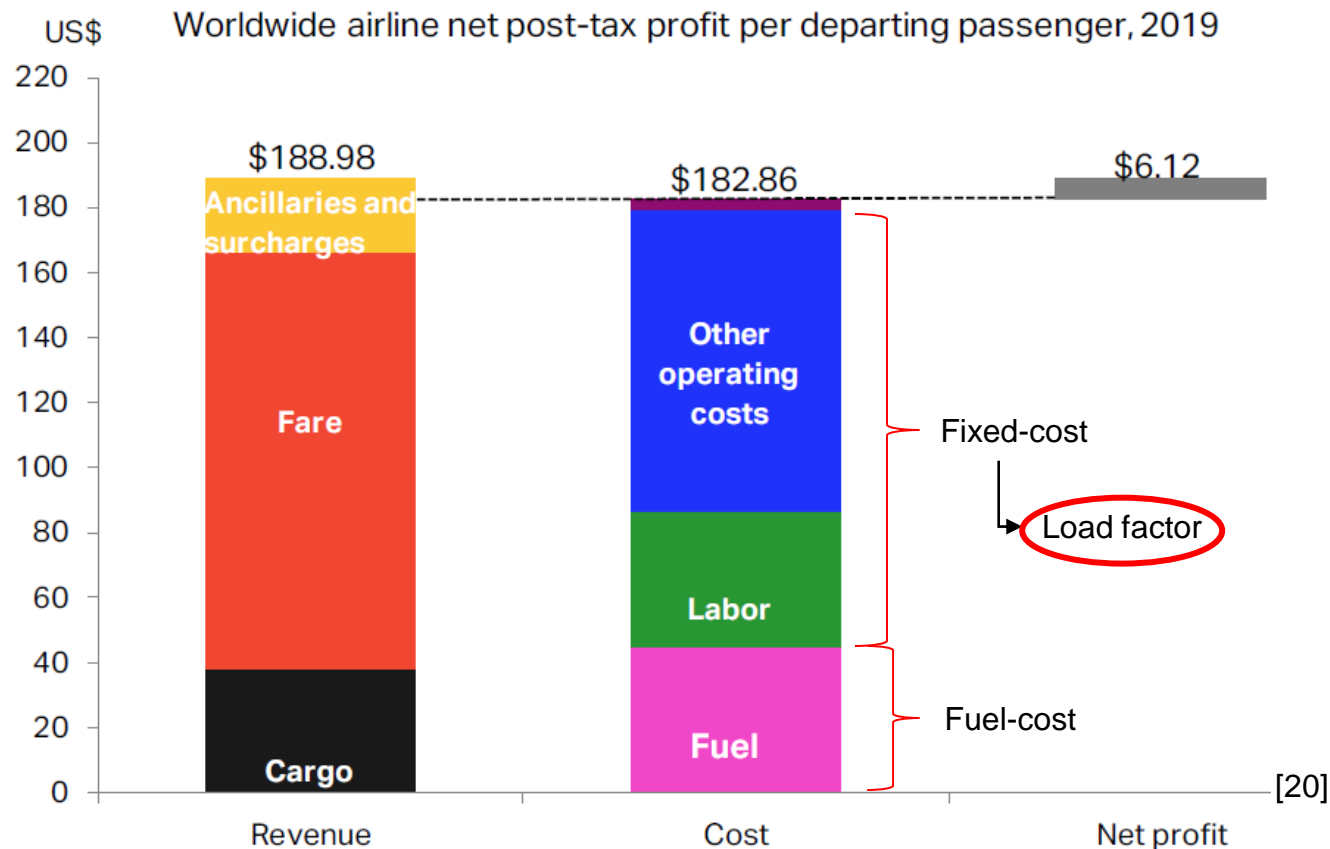


Berlin-New York : edge of space flight at 25km altitude



[19]

Fuel economy



Source: IATA Economics - Economic Performance of the Airline Industry, Mid-Year 2019

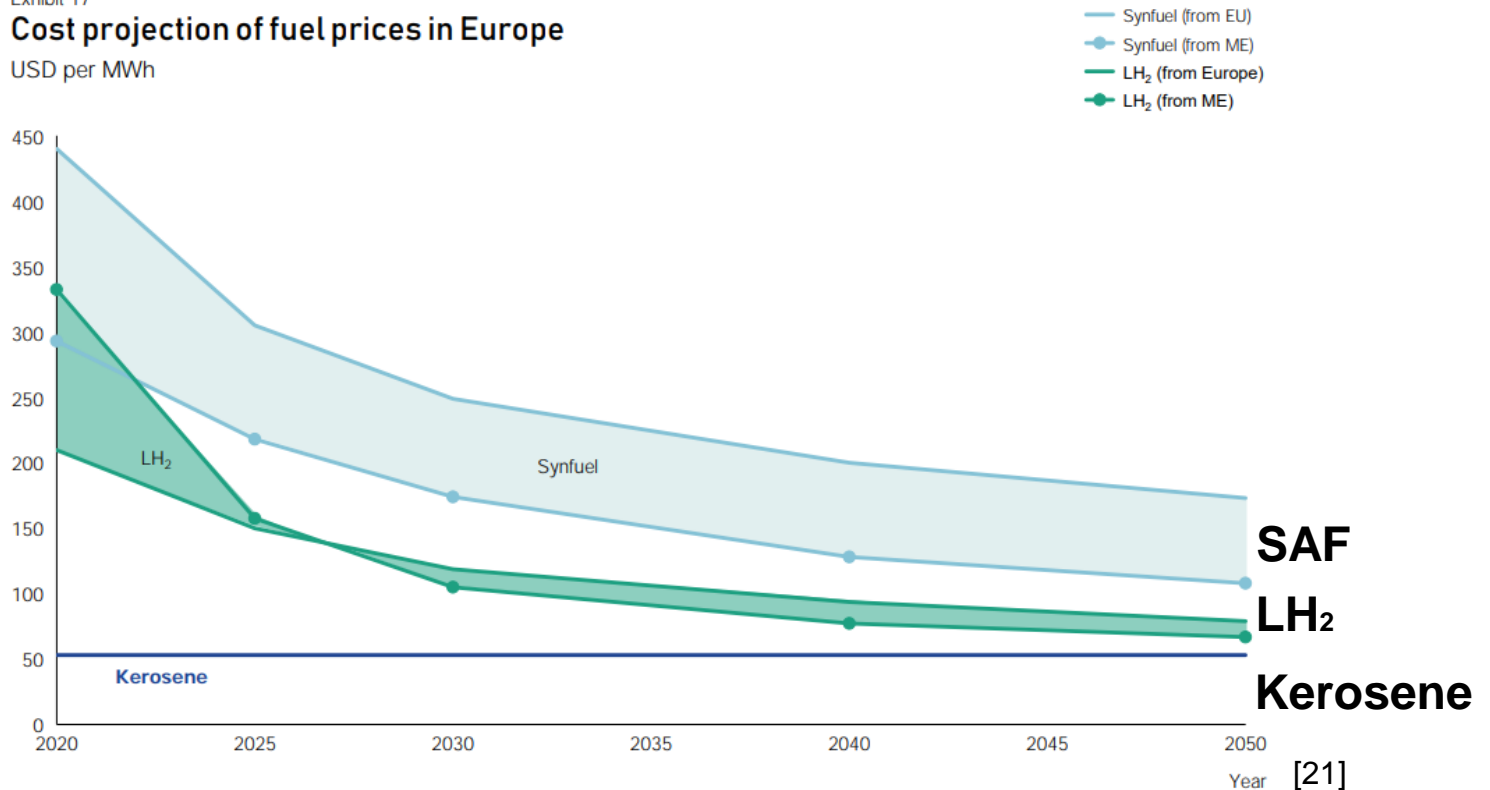
DLRK2022 | H.M.Lowhim | A new propulsion architecture for a hydrogen aircraft

Fuel economy

Exhibit 17

Cost projection of fuel prices in Europe

USD per MWh



Fuel Economy: Berlin-New York in 2050

Parameter	Unit/Symbol	Kerosene	SAF	H ₂ sub	H ₂ super
€/kg	€/kg	0.60	1.56	2.00	2.00
Energy	MWh	0.4	0.4	0.8	0.9
Frequency	flight/day	1	1	1	2
Fixed cost	€	140.86	140.86	140.86	93.91
Fuel cost	€	45.00	117.00	111.00	114.00
Ticket	€	185.86	257.86	251.86	207.91

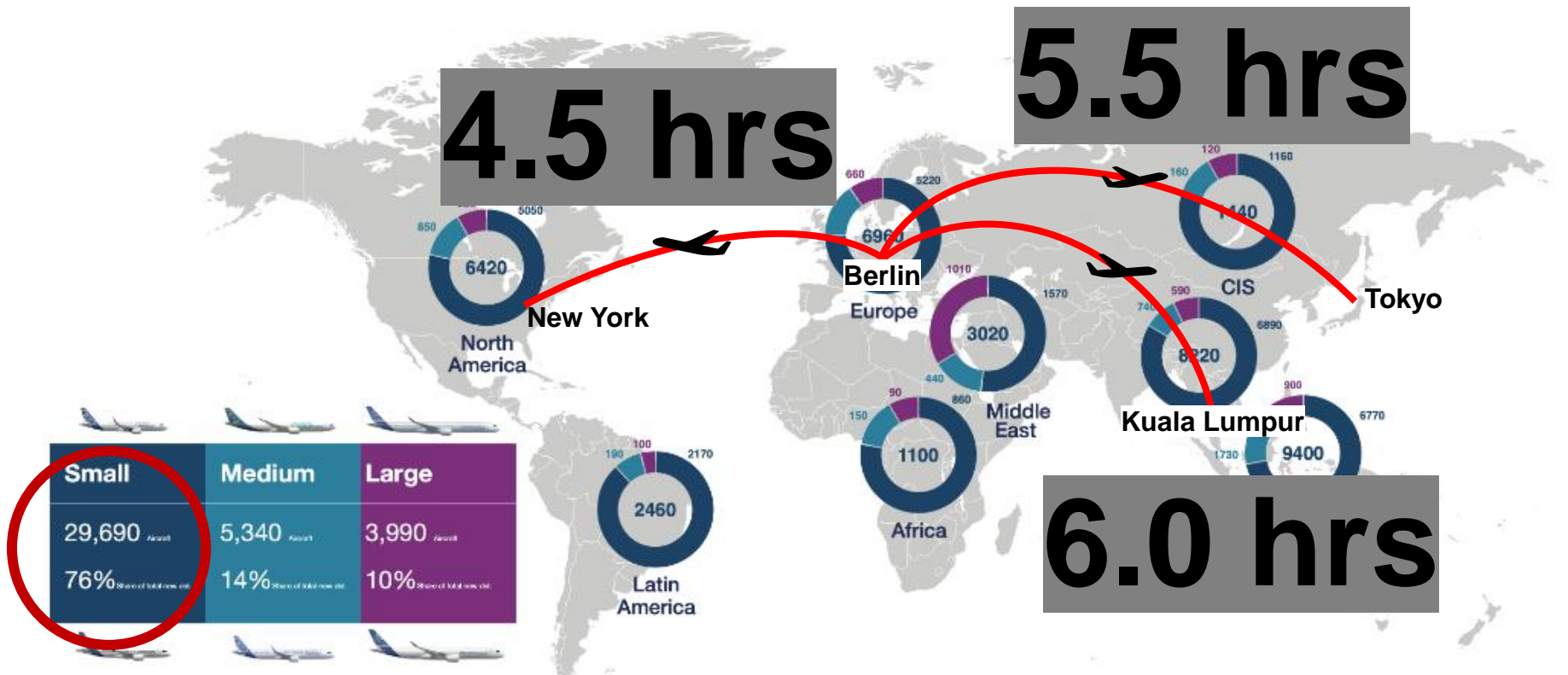
*tax and profit are not included

Air passenger: travel in half of the time

4.5 hrs

5.5 hrs

6.0 hrs



Commercial Aircraft demand 2021-2040

AIRBUS



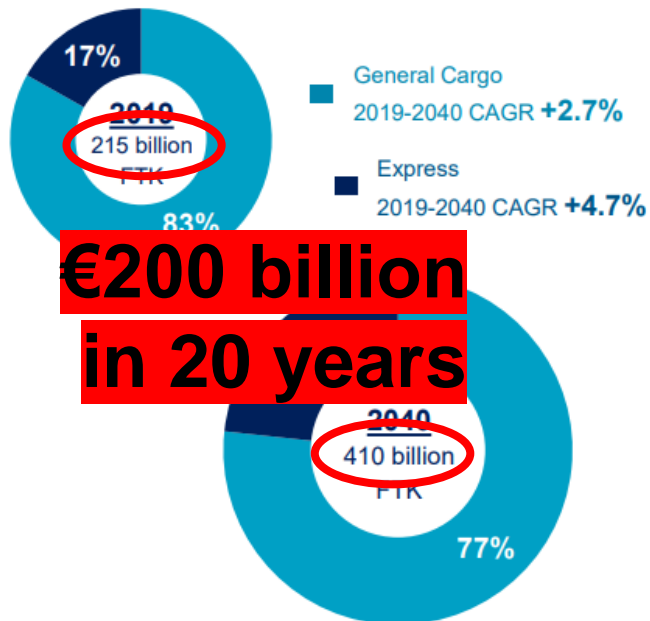
A new era of supersonic flight



0% carbon emissions at M2.0

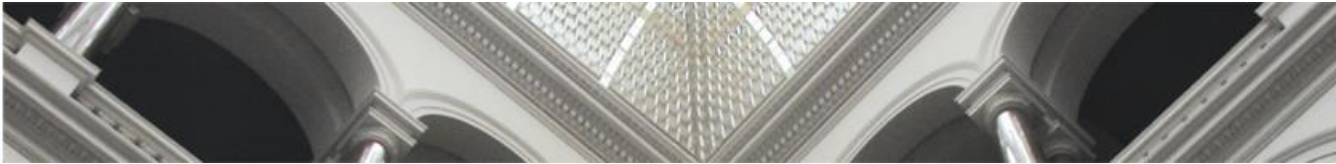
Air cargo: Bulky but cheap

World air cargo traffic



[24]

AIRBUS [23]



Summary

GI 0.8

Range

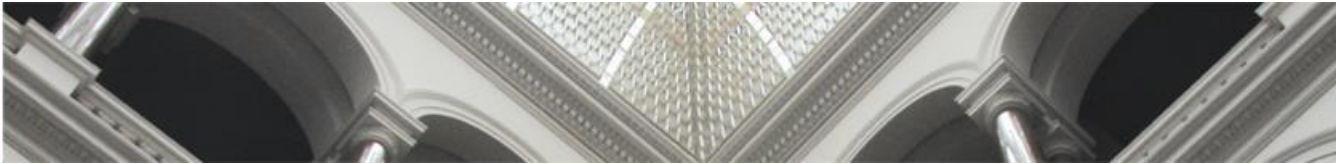
**9800
km**

Time

**4.5
hrs**

Mass

**31
ton**



Key-point

1. A **high-speed** is the solution to get a high **GI** ($GI > 0.5$)
2. A **high GI** is a key to store the LH2 within the wings and fuselage
3. **APU** will play a vital role for cryogenic management of hydrogen aircraft

GI = Gravimetric Index
APU = Auxiliary Power Unit



Key-point

4. Usability of supersonic LH₂ is **98%**

5. Transpiration cooling is a key technology for **oxygen-rich** of a H₂ gas turbine

6. Market entry time is 10-15 years (**2035**)

Did you know?

Vogel-1 propulsion architecture is a spin-off from Ariane 6 ULPM

Ariane Group contact person
Daniel Just
Daniel.just@ariane.group



Thank you!
Q&A



Main formulas

1. Braguet range equation

$$\Delta R = I_{sp} V \frac{L}{D} \ln \frac{m_0}{m_f}$$

2. Thermal exchange

$$Q = mc_p \Delta T$$

$$Q = \varepsilon \sigma A T^4$$

$$Q = -k \nabla T$$

$$Q = h A \Delta T$$

3. APU power

$$P = \frac{\Delta P \dot{m}}{\rho}$$

4. Heat exchanger

$$\Delta h = c_p \Delta T$$

5. Density

$$\rho = \frac{m}{v}$$

6. Gravimetrix index

$$GI = \frac{m_{fuel}}{m_{fuel} + tank + fuel\ system}$$

7. Cabin length

$$L_{Cabin} = S_P \left| \frac{PAX}{S_A} \right| + L_{galley}$$



Image index

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2. <https://aircraft.airbus.com/en/aircraft/a320/a321xlr>
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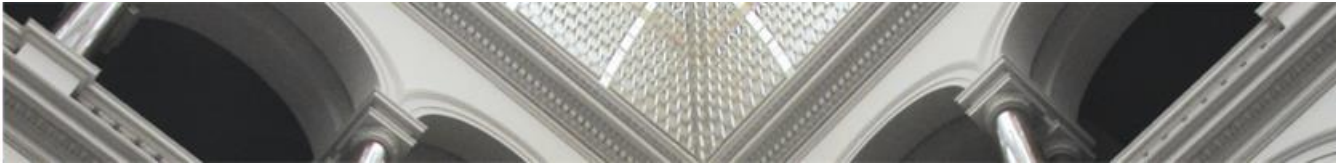
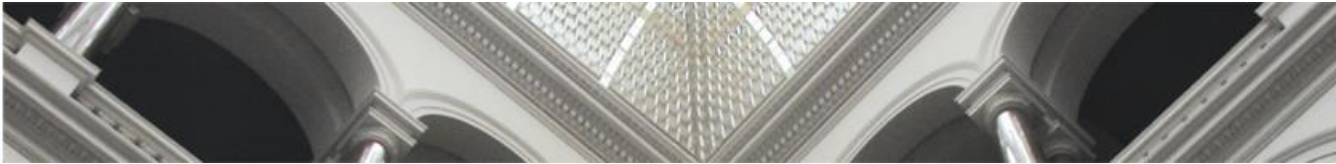


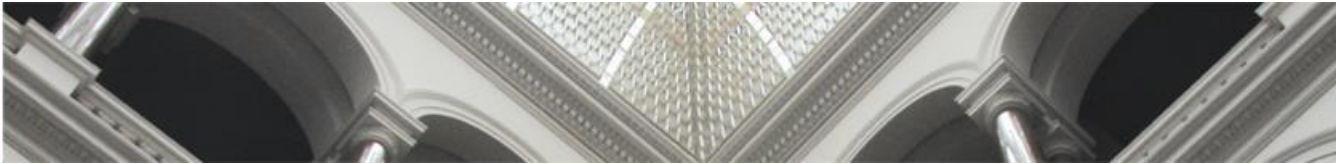
Image index

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3. Thermal Control of the Cryogenic Upper Stage of Ariane 6, Ariane Group Bremen , 49th International Conference on Environmental Systems, 2019



Glossary

1. MLI = Multi Layer Insulator
2. NGS = Nitrogen Generation System
3. RAT = Ram Air Turbine
4. APU = Auxiliary Power Unit
5. GH2 = Gaseous Hydrogen
6. LH2 = Liquid Hydrogen
7. LN2 = Liquid Nitrogen
8. GN2 = Gaseous Nitrogen
9. SH2 = Saturated Hydrogen
10. EHC = Electrochemical Hydrogen Compressor
11. P-R = Pressure Regulator
12. F-R = Flow Regulator
13. EH = Electrical Heater
14. RFTC = Radial Flow Turbine Compressor
15. AFT = Axial Flow Turbine